

## GLASAIR II-S TD ADVANCE NOTICE OF REVISION

**Page C-42**, FIGURE (C-31) and first sentence in Step J-2: Change "MSP-42" to "700-0042-001".

**Page C-90**, Steps T-3 and T-4: Delete references to static lines in the wings. The static system for the Glasair now uses fuselage mounted ports.

**Page C-114**, second note: Change "H ribs" to "I ribs".

**Page C-174**, FIGURE (C-120): Change "Cabosil/Resin Mixture" to "Mill-Fiber/Resin Mixture".

**Pages E-12 through E-14**: Since the shape of the aft vertical fin shearweb has been changed, as shown in FIGURE (E-5A), the installation of the foam wedges will change. Build up the right side of the shearweb with additional pieces of foam to make a flat surface for installing the wedges.

**Page G-289**, first sentence: Use shielded wire on all **magneto wire connections**, and ground to the nearest ground (-) bus.

**Page G-326**, FIGURE (G-185): Change "Header Tank Fuel Pickup Fitting" to "Bulkhead Fitting."

  
**STODDARD-HAMILTON**  
AIRCRAFT, INCORPORATED

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